



Space Dynamics

LABORATORY

Utah State University Research Foundation

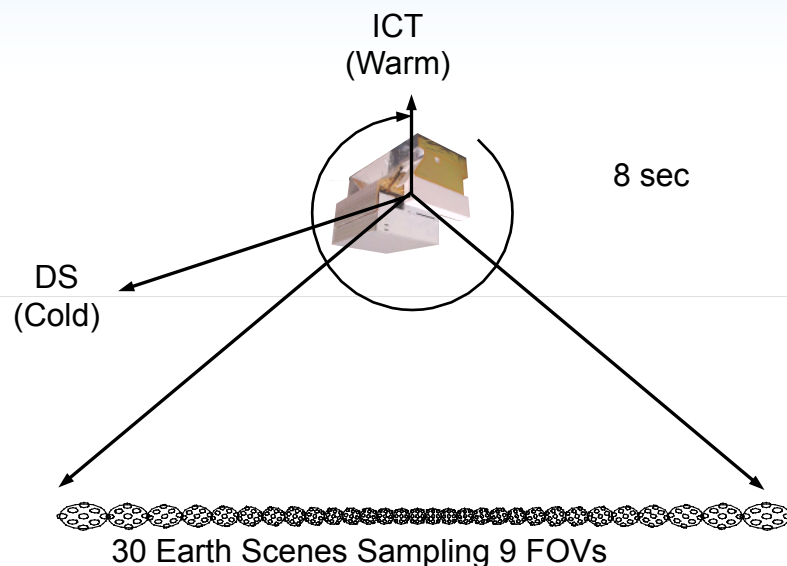
CrIS Internal Target Emissivity Check From Day in the Life Test Data NASA Souder Science Team Meeting

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CrIS Sensor On The NPP Satellite

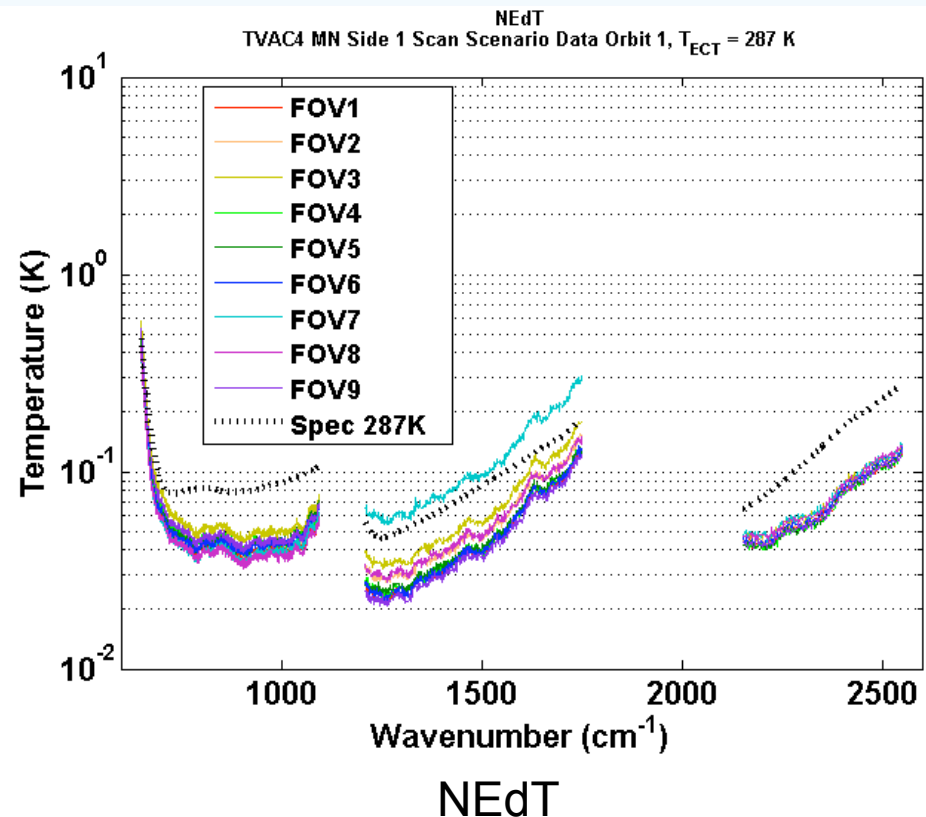
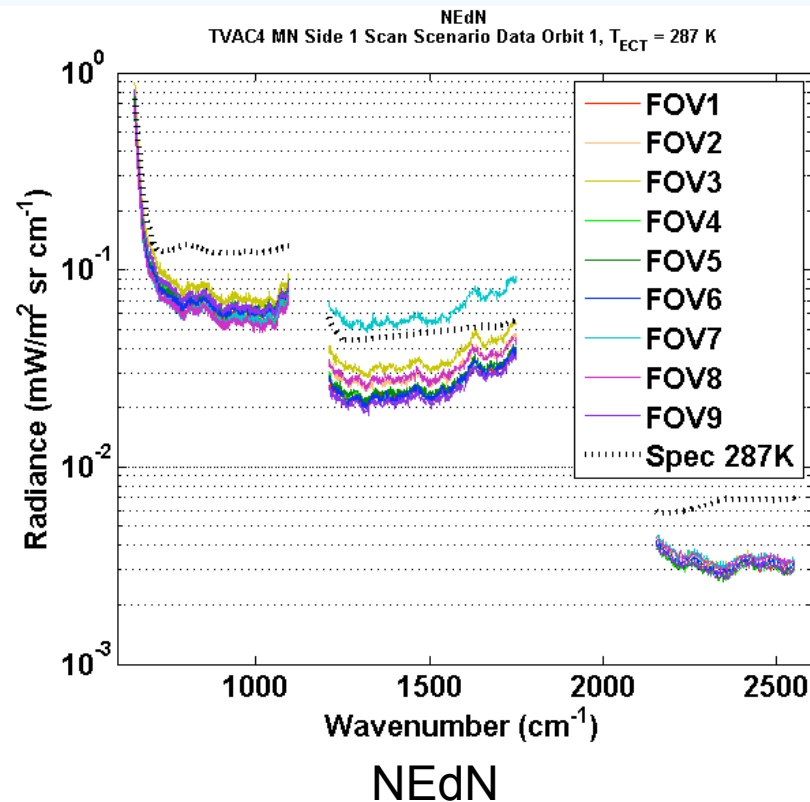
- ▶ 8-second scans
- ▶ 30 Earth locations
 - 9 FOVs per location
 - 3 spectral bands per FOV
 - LWIR 650-1095 cm^{-1} , resolution: 0.625 cm^{-1}
 - MWIR 1210-1750 cm^{-1} , resolution: 1.25 cm^{-1}
 - SWIR 2155-2550 cm^{-1} , resolution: 2.5 cm^{-1}
- ▶ Two calibration views per scan
 - Internal Calibration Target (ICT) — Warm (ambient temperature)
 - Deep Space (DS) — Cold
 - Separate calibration for two interferometer scan directions
- ▶ CrIS has completed thermal vacuum testing and is now undergoing spacecraft integration



Day in The Life Test

- ▶ Also known as the Scan Scenario test
- ▶ Scene scan module and electronic box temperatures driven through 3 simulated orbits
- ▶ Voltage varied representative of on-orbit bus voltage
- ▶ Primary purpose was to provide a flight like data set to test software
- ▶ Analysis of TVAC3 data showed a problem with the ICT temperature sensing electronics
- ▶ TVAC4 performed to validate modifications to the electronics
- ▶ Opportunity to try out Cal/Val type techniques

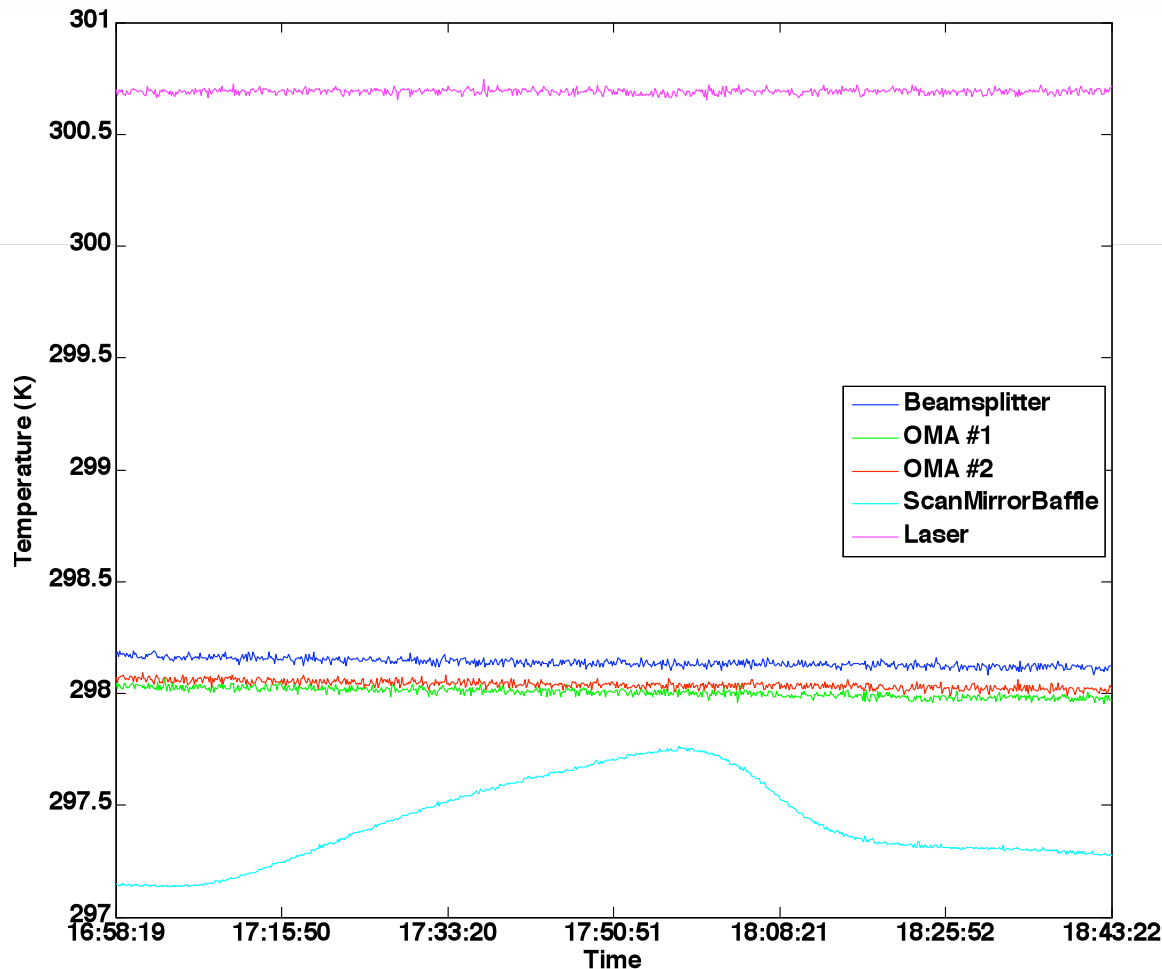
Noise Performance During Scan Scenario



- Measured brightness temperature variations were small compared to random noise
- Substantial averaging was needed to see any radiance errors

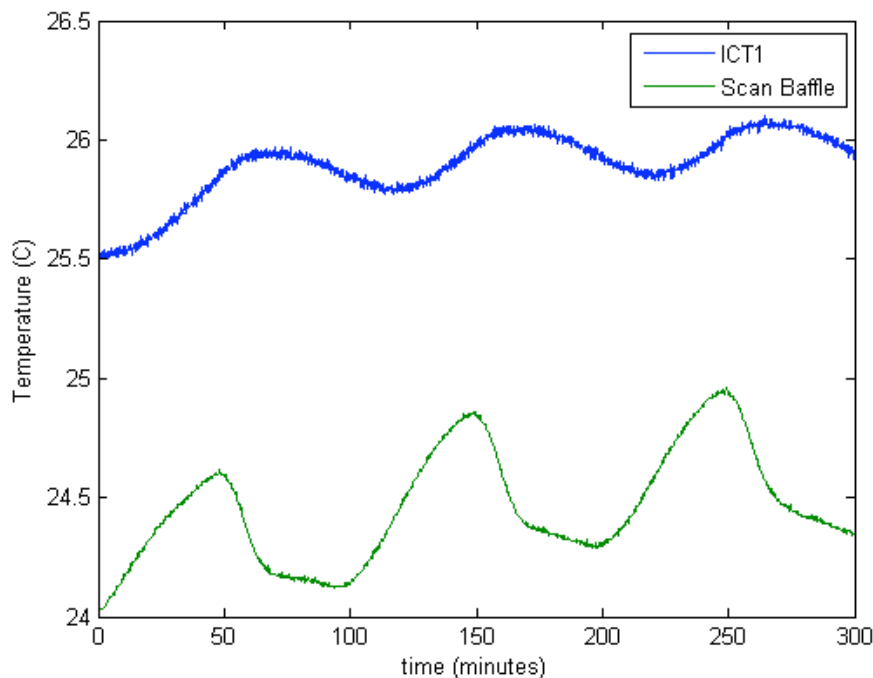
Temperature of Sensor Components Over One Simulated Orbit

MN Scan Scenario Temps

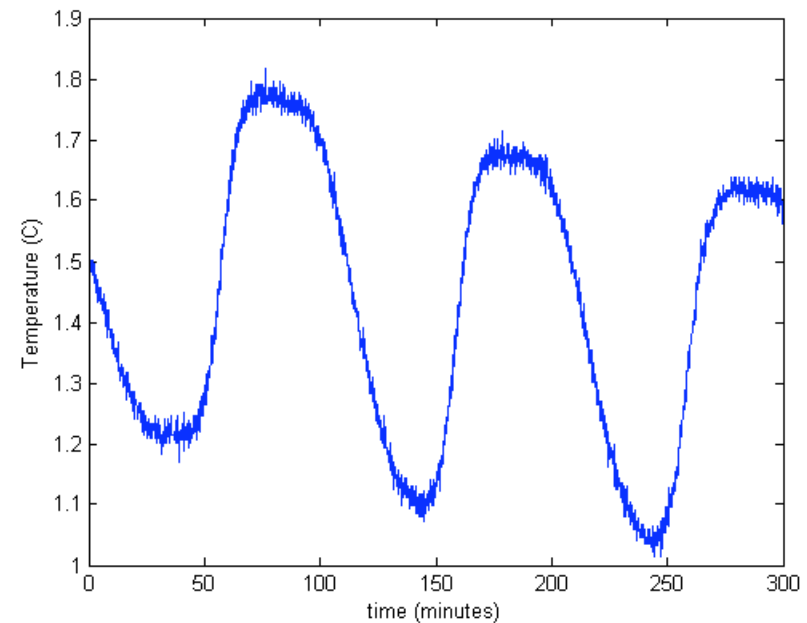


- ▶ Radiance from the environment of ICT can be reflected into the interferometer
- ▶ Most system components are very stable thermally over an orbit
- ▶ Scan baffle only system component with a view to the ICT that has significant temperature variation

Temperature of the ICT and Scan Baffle



Temperature difference between ICT1 and Scan Baffle

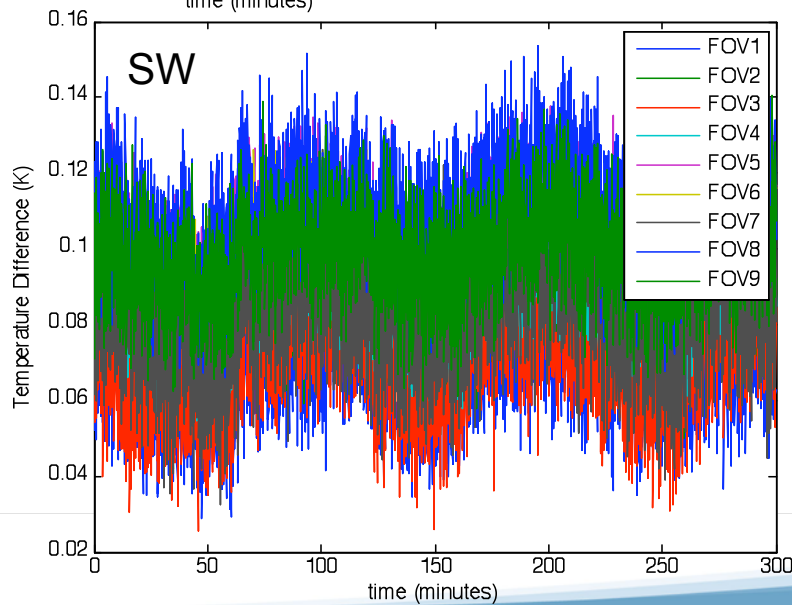
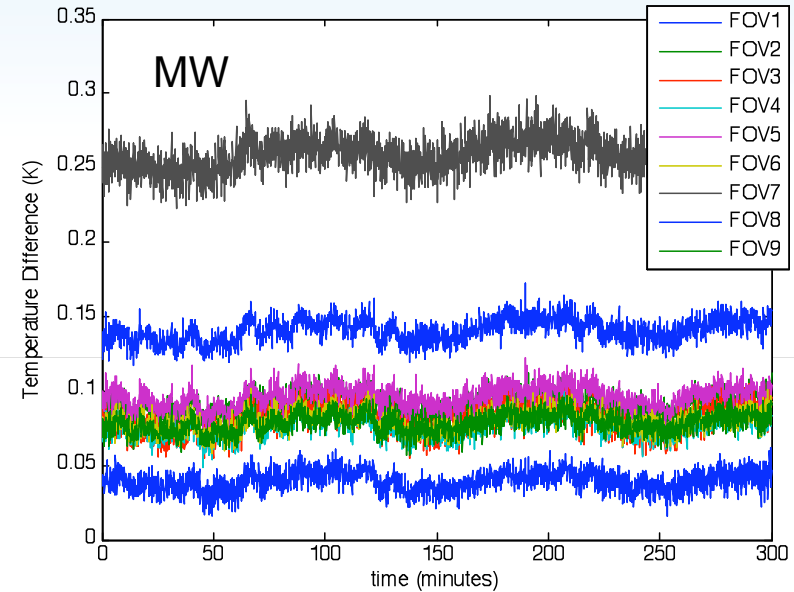
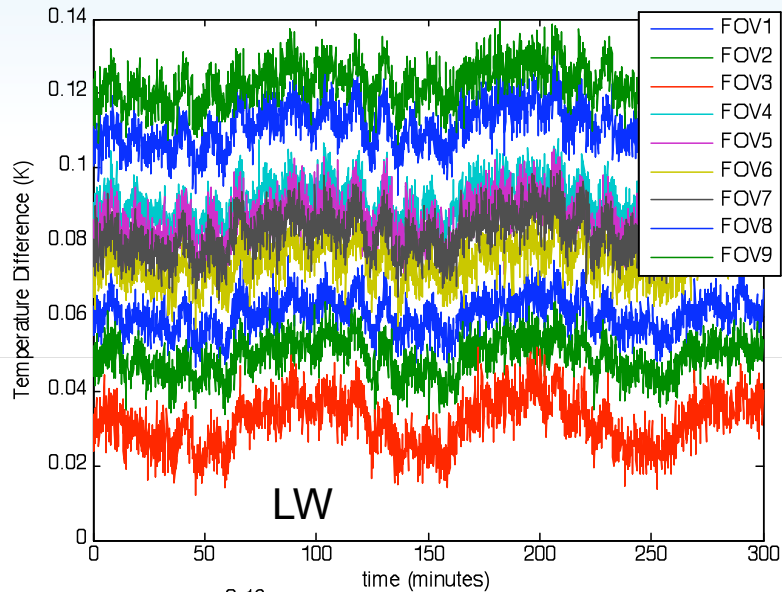


- Errors caused by reflected radiance from the scan baffle would be expected to correlate with temperature difference between the ICT and the scan baffle

Radiometric Time History of Scan Scenario

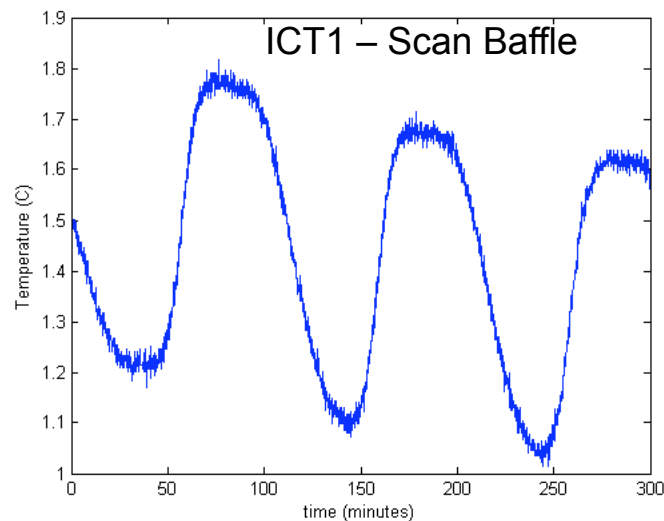
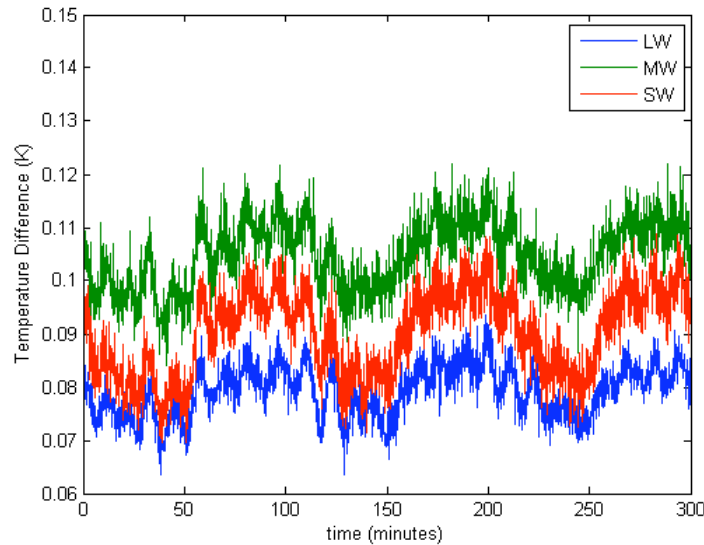
- ▶ Spectra were spectrally averaged then plotted as a time history (spectral content averaged to give a single point for each spectra)
- ▶ The source was a constant temperature 287K ECT
- ▶ Variation of the CrIS measured brightness temperature with sensor temperature represents a radiance error
- ▶ ITT SDR_Generator version 2.18 with no ILS correction
- ▶ Nonlinearity correction coefficients taken from TVAC3
- ▶ Some FOV to FOV spread is also caused by temperature gradients in the ECT

Spectrally Averaged Time Histories



TVAC4 Scan Scenario Side 1

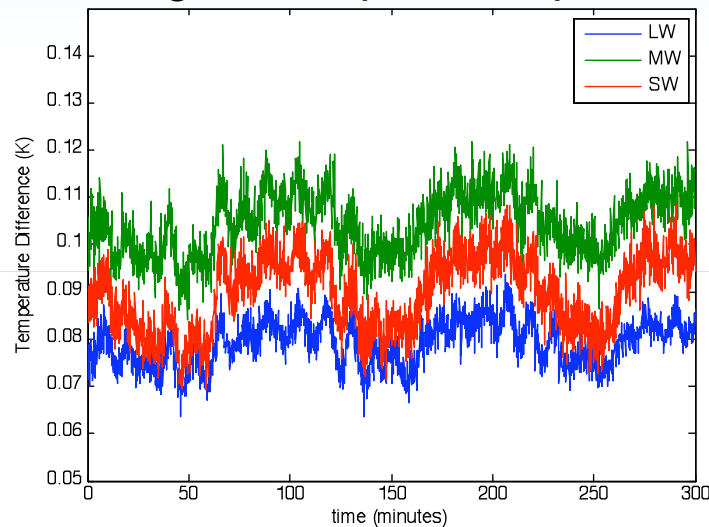
Radiance Errors Track ICT Scan Baffle Temperature Difference



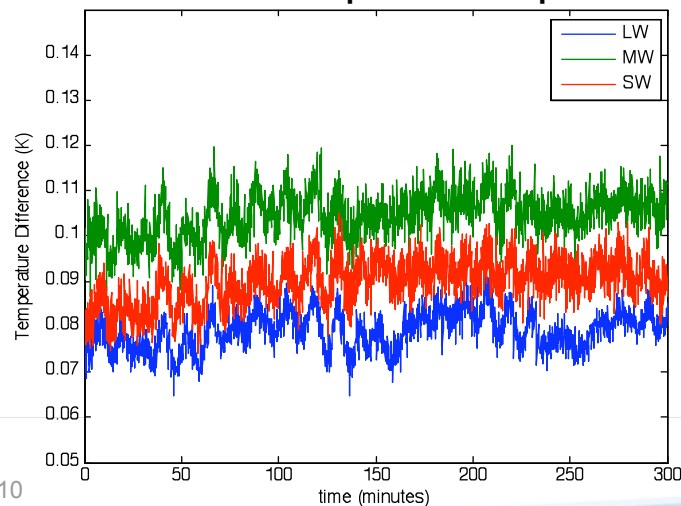
- ▶ All FOV averaged together
- ▶ Indication of radiance error being caused by reflections from the ICT
- ▶ Amplitude of radiance error for different bands follows ICT emissivity pattern
- ▶ ICT emissivity in SW is lowest so higher radiance error is expected
- ▶ Phase of radiance error tracks ICT minus scan baffle temperature

Radiance Error Nulled by Modifying the Scan Baffle Temperature

Original temperature profile

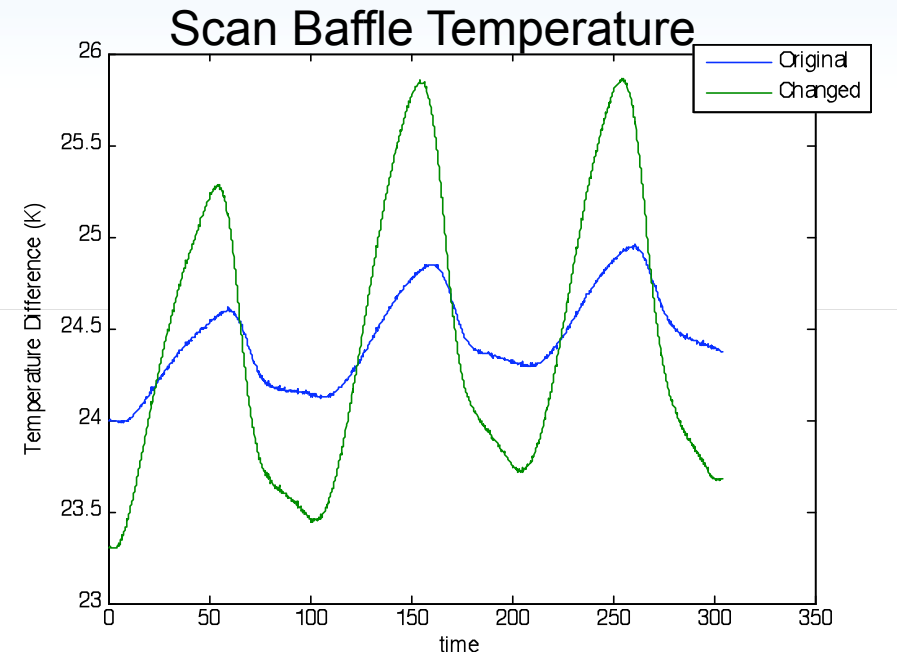
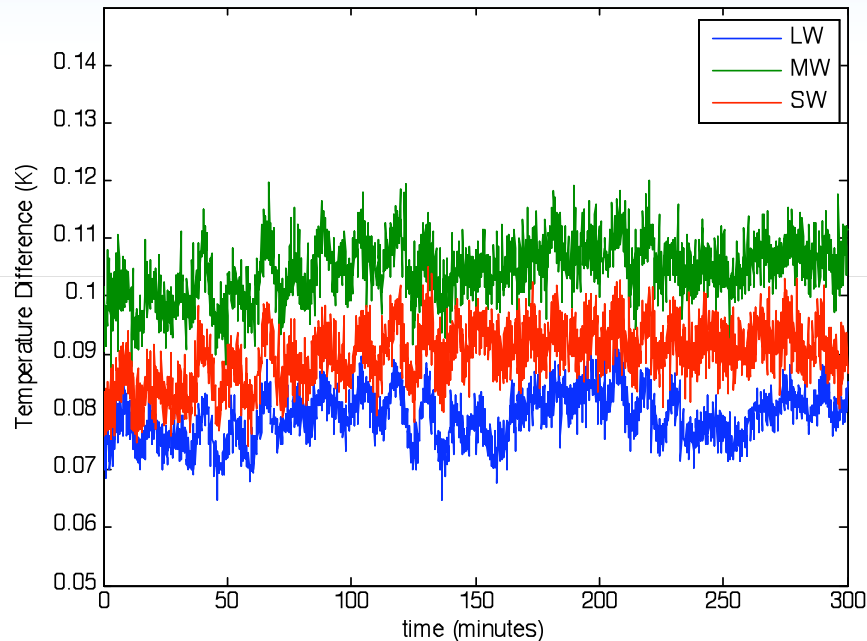


Modified temperature profile



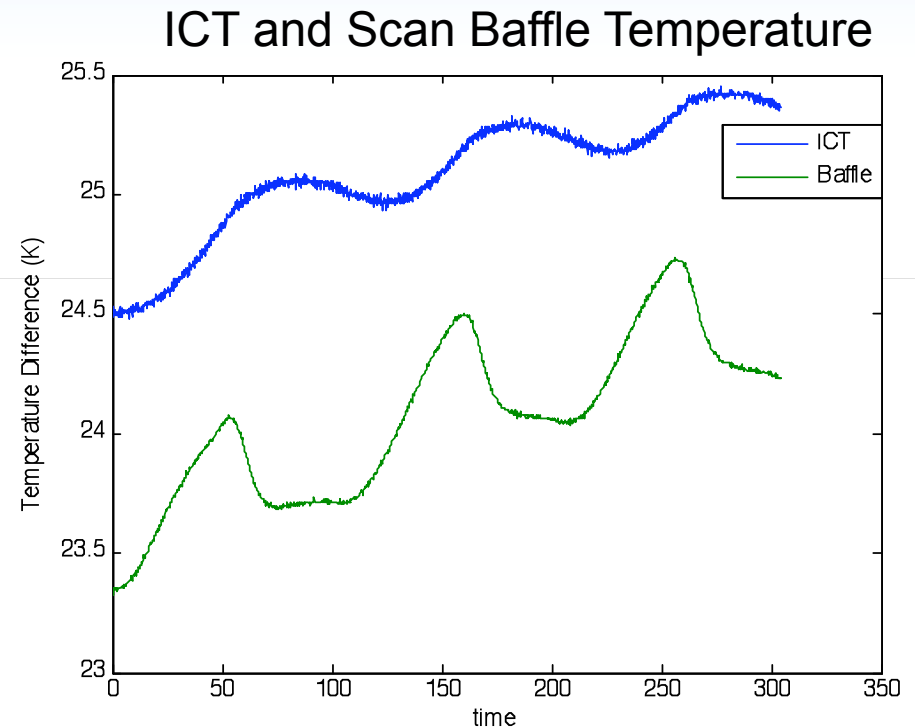
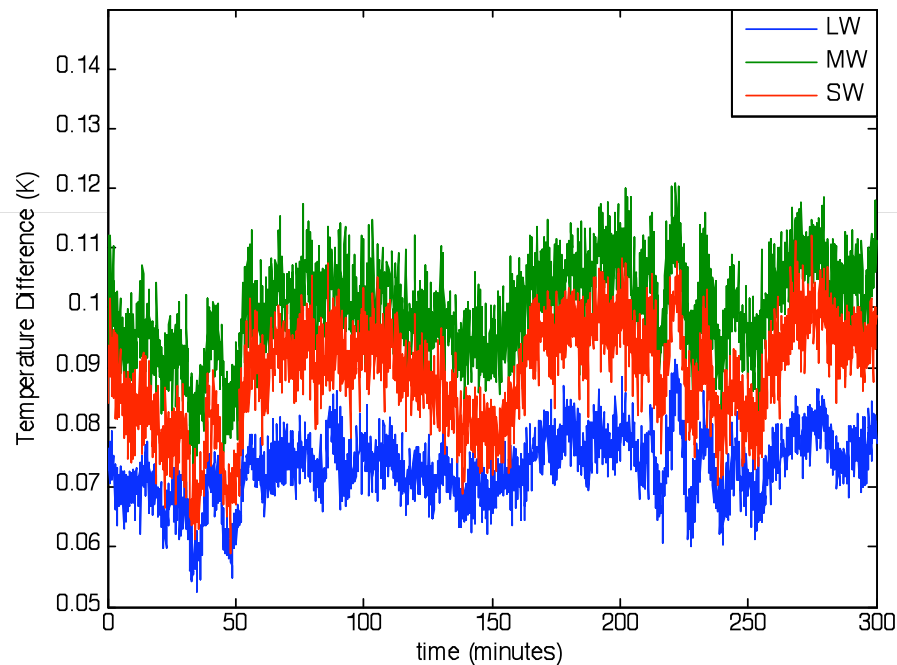
- ▶ Scan baffle temperature sensor located on base of baffle insulated from temperature extremes
- ▶ Portion of scan baffle viewed by ICT is likely to have larger temperature extremes and change temperature faster than temperature sensor
- ▶ Scan baffle temperature profile modified and radiance recalculated

Modified Scan Baffle Temperature Profile Reduces Orbital Variation



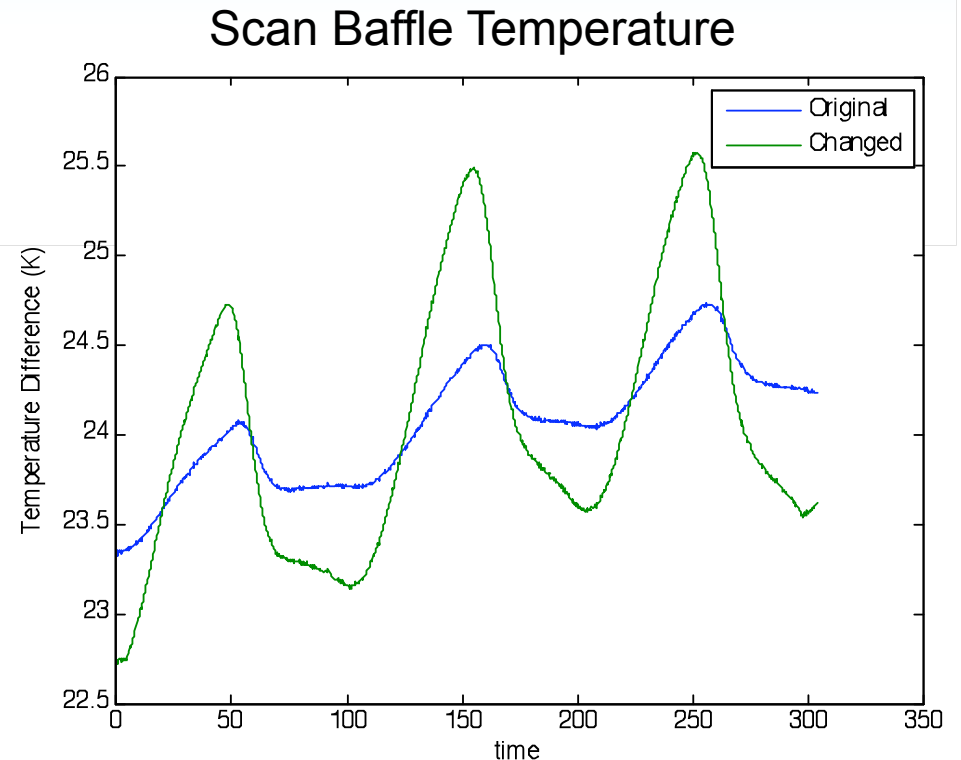
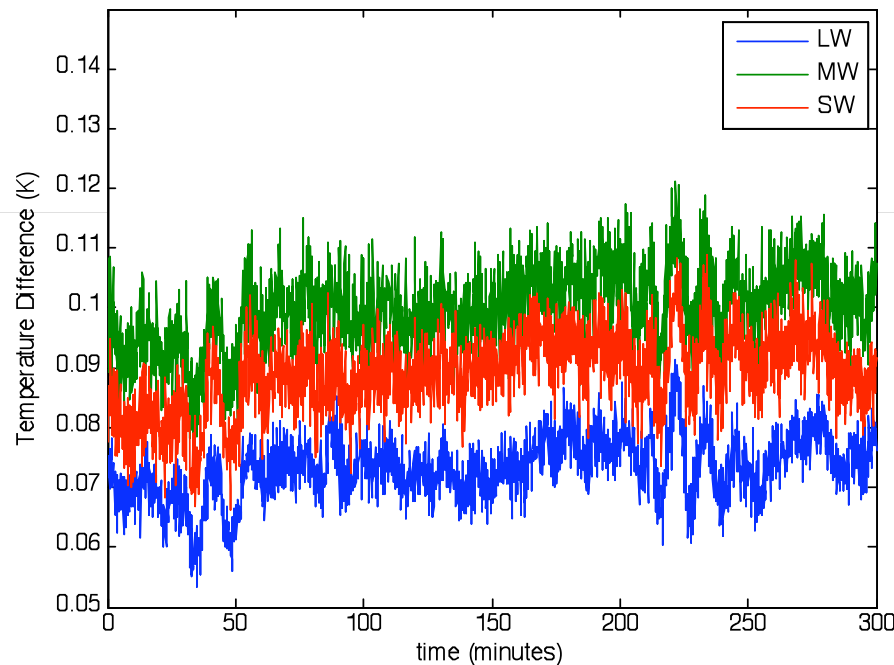
- ▶ AC part of scan baffle temperature profile increased by 1.03 K and phase adjusted to give a 6 minute time advance
- ▶ Correction for the LW and MW not complete
- ▶ ICT emissivity used in environmental model for the LW and MW too large

Side 2 Scan Scenario Results



- ▶ Side 2 Scan Scenario results similar
- ▶ A little more ECT temperature variations
- ▶ ECT temperature variations are the same magnitude in each band

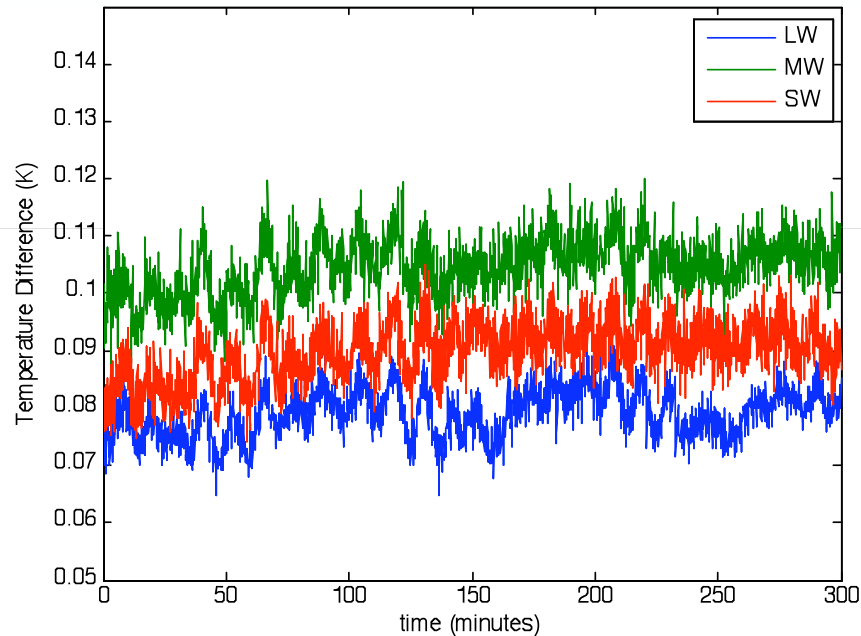
Side 2 With Modified Scan Baffle Temperature



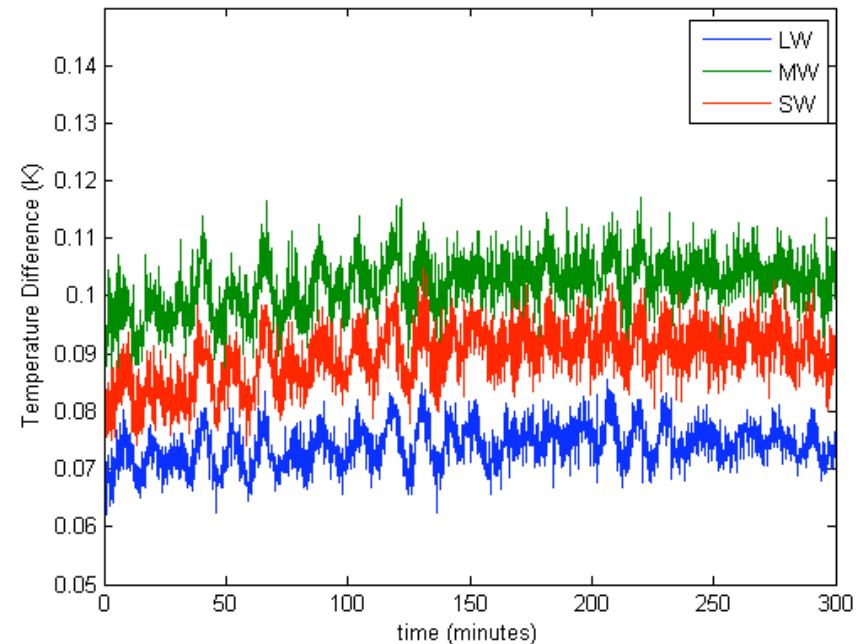
- AC part of scan baffle temperature profile increased by 1.03 K and phase adjusted to give a 6 minute time advance

Modifying ICT Emissivity Reduces Radiance Error

Modified scan baffle temperature



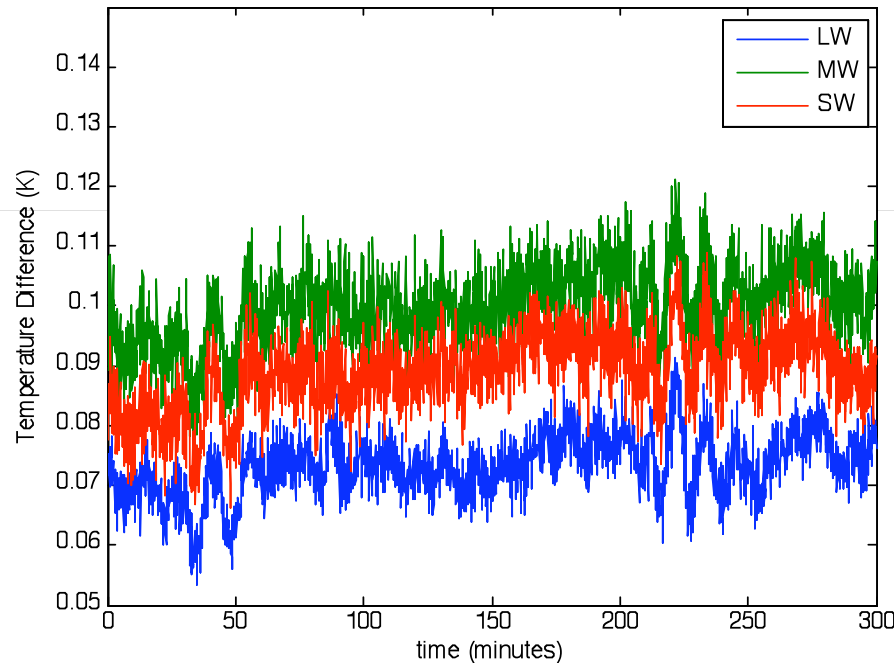
Also modifying ICT emissivity



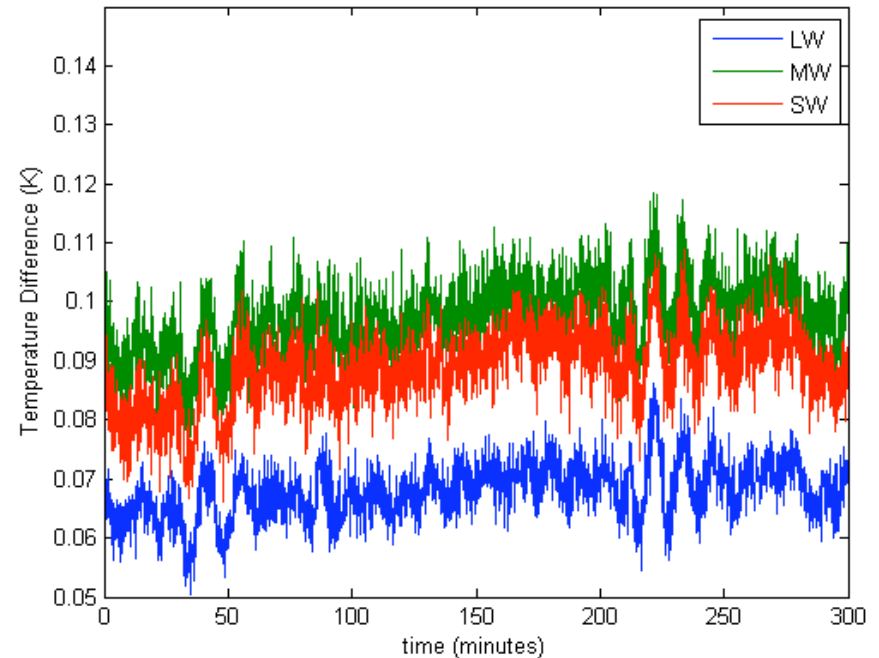
- Modifying the ICT emissivity as well as the scan baffle temperature profile reduces radiance error
- This is a band to band relative emissivity check not an absolute measurement

Side 2 Results are Similar

Modified scan baffle temperature

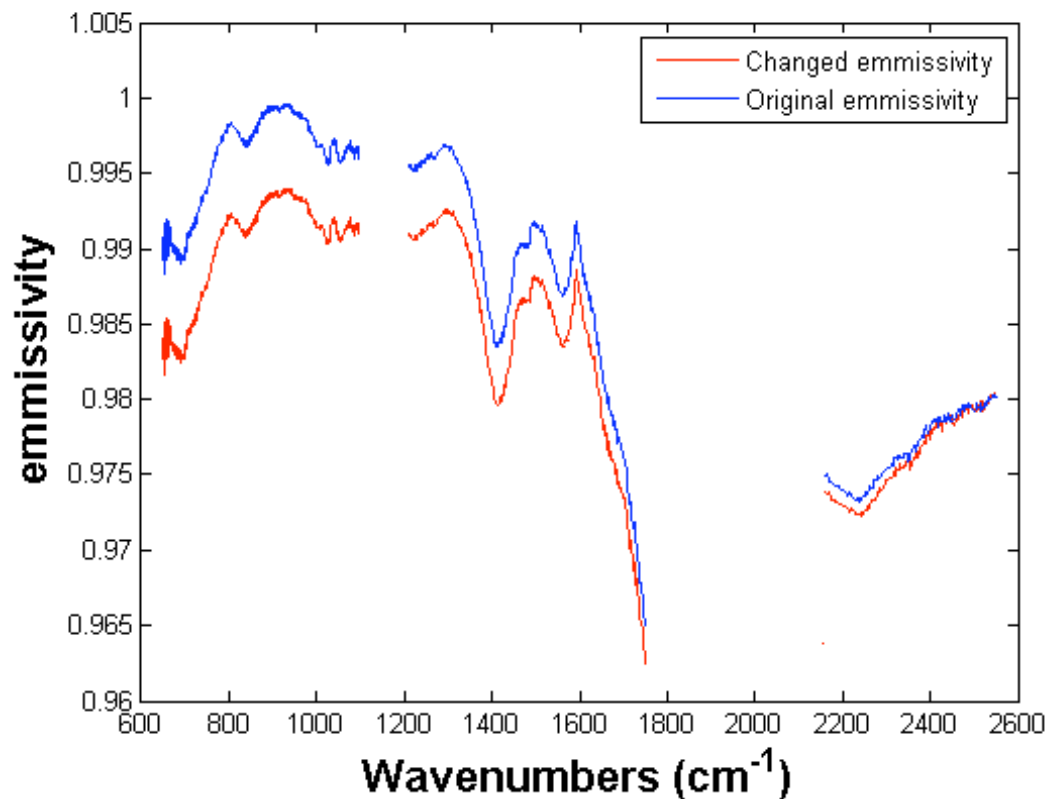


Also modifying ICT emissivity



- Modifying the ICT emissivity as well as the scan baffle temperature profile for side 2 produces similar results
- There is a little more ECT temperature variation for side 2

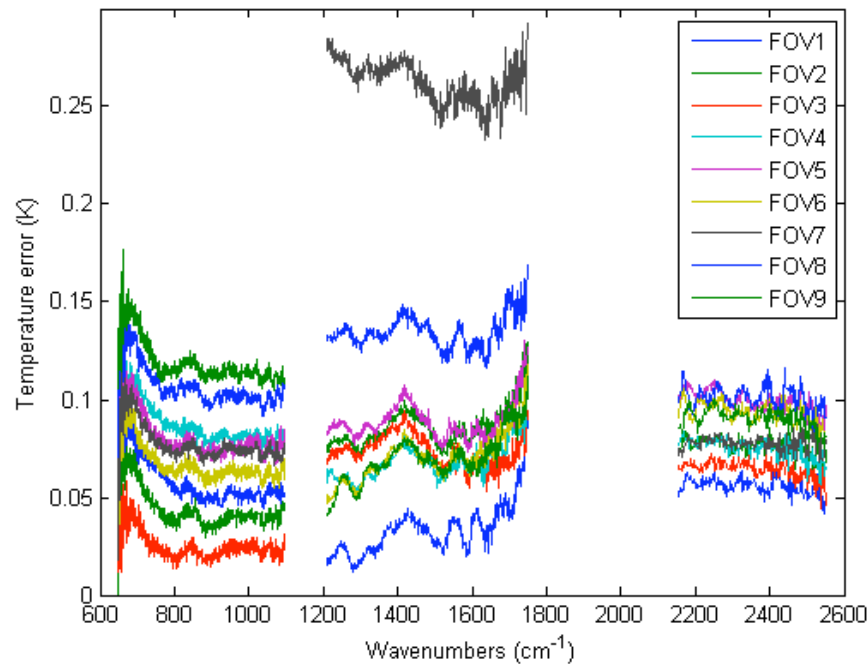
Modified Emmissivity Reduces Radiance Error



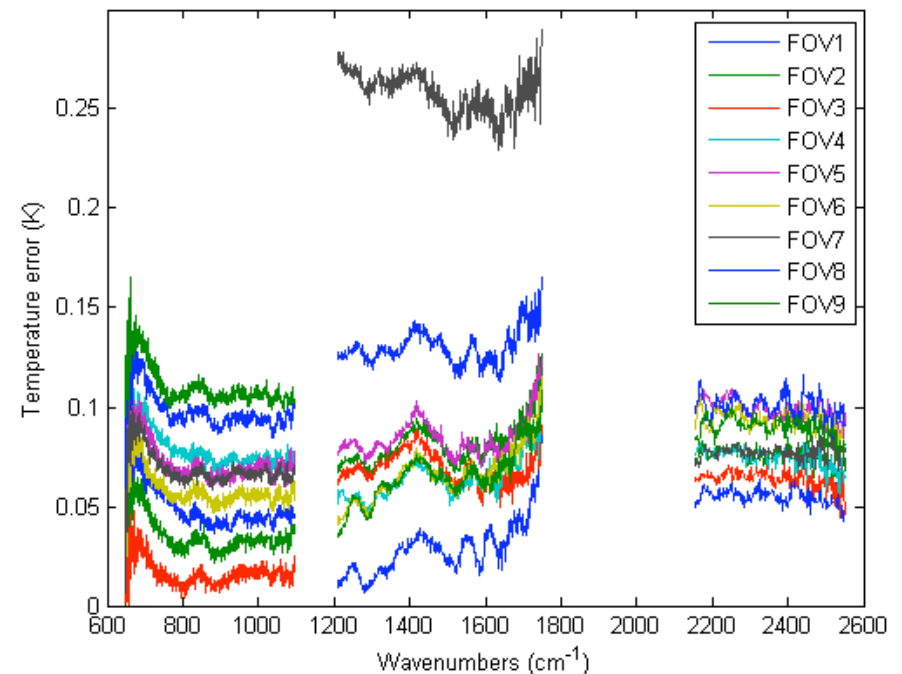
- Spectral shape of ICT emissivity determined by ITT using CrIS measurements
- ICT emissivity anchor point set in the SW band using radiometer measurement
- Engineering packet contains ICT emissivity
- Modification to ICT emissivity consisted of linear reduction of 0.0067 at the longwave end of band

Modifying ICT Emissivity Did Not Significantly Affect Radiometer Uncertainty

Original Emissivity

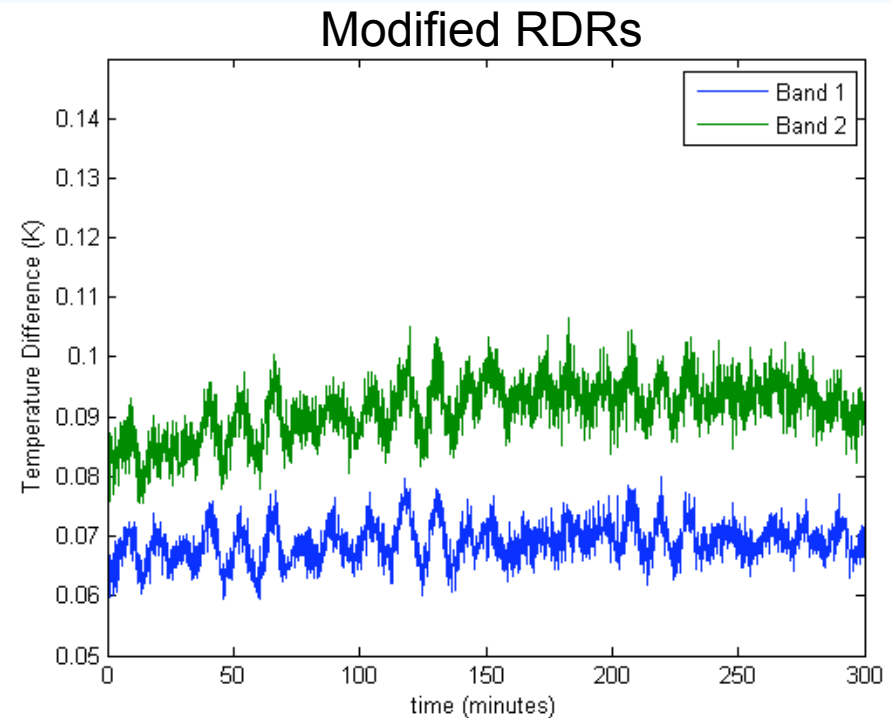
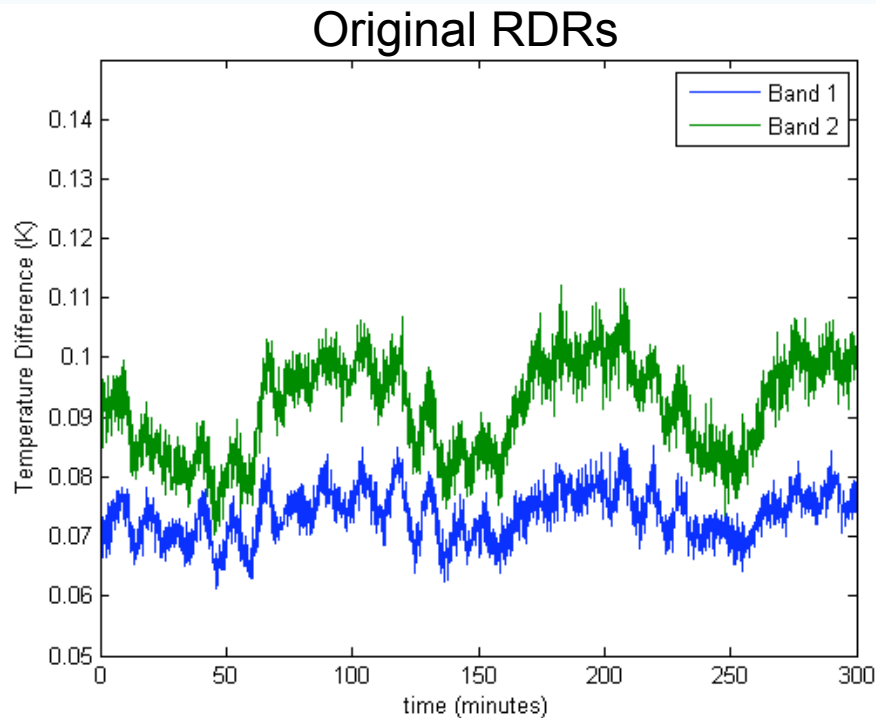


Modified Emissivity



- Emissivity modified by 0.0067 at end of LW band
- Not using latest nonlinearity a_2 coefficients
- No Scan baffle offset

Comparison of Bands with Different Emissivities



- ▶ Band 1: 860 – 1000 cm^{-1} (high emissivity)
- ▶ Band 2: 2155 -2340 cm^{-1} (lower emissivity)
- ▶ Scan baffle and emissivity modified as in previous slides

Alternative Approach: Solve for Scan Baffle Offset Temperature

- ▶ Many errors cancel to first order (especially relative errors)
 - ECT temperature
 - Nonlinearity
 - Radiance from component with stable temperatures
 - ICT temperature (non-time dependent)
- ▶ Assume remaining radiance error are caused by scan baffle temperature offset
- ▶ Sensitive to
 - Emissivity knowledge of ICT and ECT
 - ICT time dependent temperature knowledge (TVAC3 problem)
 - Environmental model errors
- ▶ Use extensive averaging to reduce noise



Linearized Version of Brightness Error

$$\begin{aligned} B_1 &= T_{ECT} + S_{SB1} T_{SBoff} + S_{ICT1} T_{ICT} \\ B_2 &= T_{ECT} + S_{SB2} T_{SBoff} + S_{ICT2} T_{ICT} \end{aligned}$$

Where:

B_1 brightness temperature for band 1

B_2 brightness temperature for band 2

T_{ECT} temperature of the ECT

T_{ICT} temperature of the ICT

T_{SBoff} temperature of the scan baffle

S_{SB1} sensitivity of band 1 to the scan baffle temperature

S_{SB2} sensitivity of band 2 to the scan baffle temperature

S_{ICT1} sensitivity of band 1 to the ICT temperature

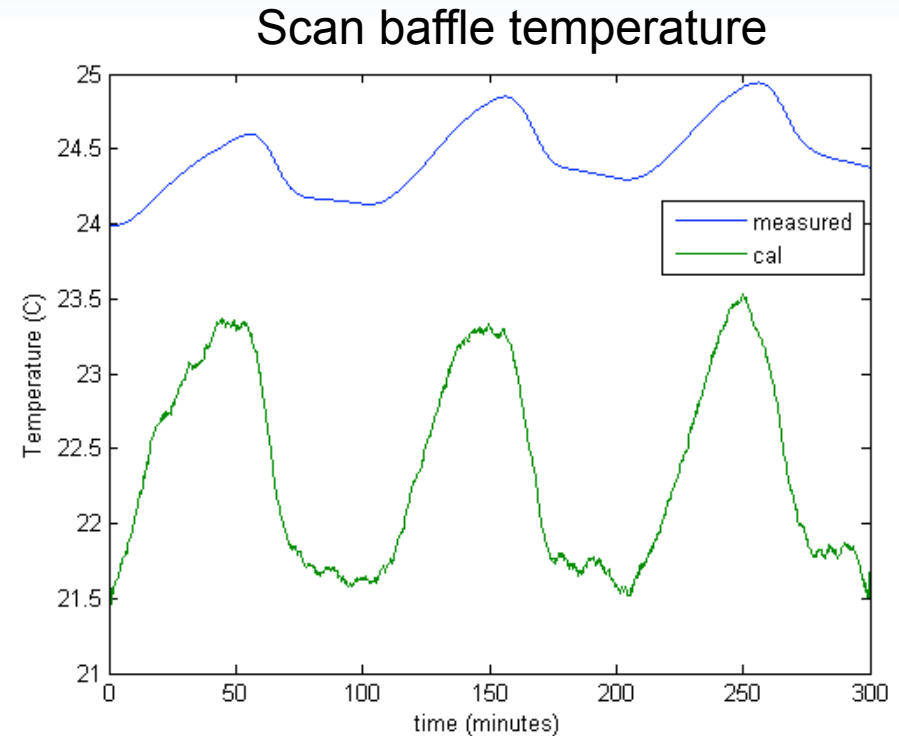
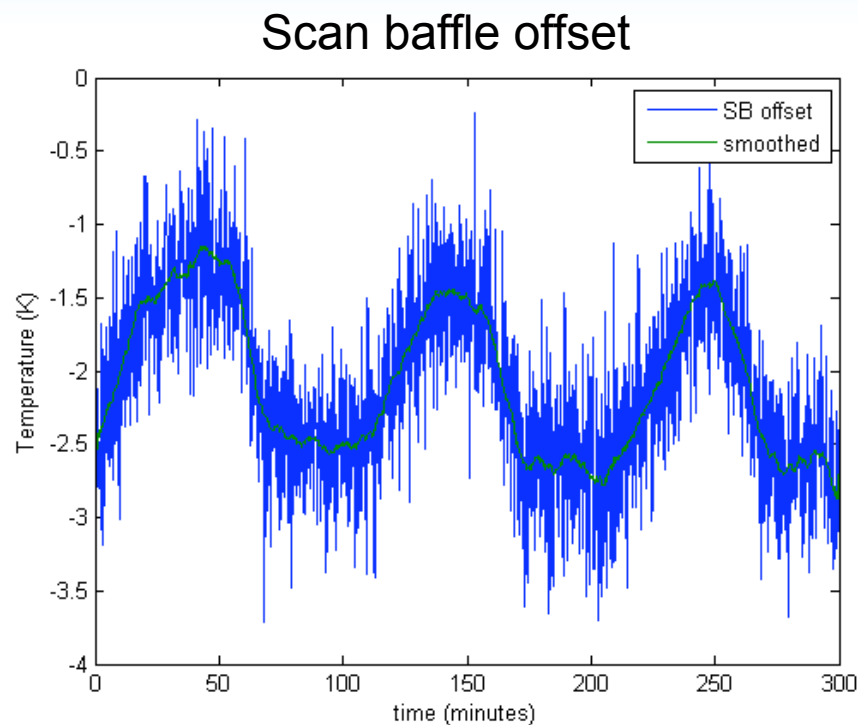
S_{ICT2} sensitivity of band 2 to the ICT temperature

$$S_{ICT1} \approx S_{ICT2}$$

Solving for the temperature of the scan baffle offset gives

$$T_{SBoff} \approx (B_1 - B_2) / (S_{SB1} - S_{SB2})$$

Calculated Scan Baffle Offset



- ▶ Calculated scan baffle temperature looks reasonable
- ▶ Absolute scan baffle offset temperature similar to ITT MN value of -2.5 K

Conclusion

- ▶ The CrIS sensor has completed thermal vacuum testing and is now being integrated with the spacecraft
- ▶ Extensive data averaging makes possible the detection of small radiance error during the scan scenario test
- ▶ Modification of the scan baffle temperature profile reduces this error
- ▶ Indication that the LW ICT emissivity in the engineering packs is slightly too high relative to the SW emissivity
- ▶ Reasonable scan baffle temperature calculated from scan scenario radiance error
- ▶ A time varying scan baffle temperature offset is planned for use on orbit

